

# **Center for Aircraft Structural Life Extension**

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*Providing Structural Integrity Technology to the Aerospace Community*



**U.S. AIR FORCE**

**AFGROW COM  
programming for  
EIFS calculations**

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Extension  
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- EIFS Assumptions and Background
  - EIFS Calculation Process and Input Data
  - Crack Size Probability Distributions
  - EIFS Calculations and Updated Guess Logic
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# EIFS Basics

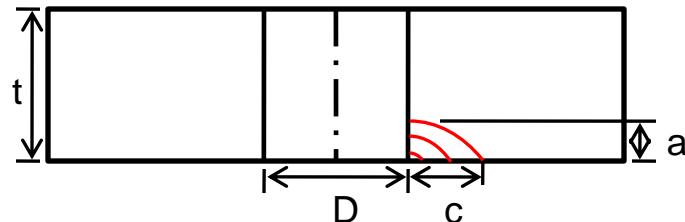
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## ■ Equivalent Initial Flaw Size

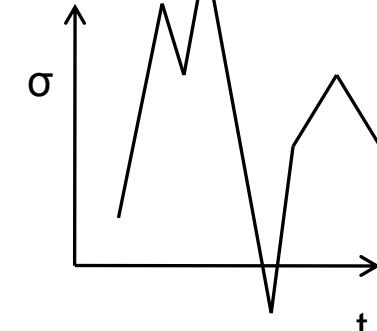
- Assumes Linear Elastic Fracture Mechanics valid
- Material crack growth properties
- Assumed loading in actual structure
- “Real-world” crack geometry and life

# EIFS Process

Real Crack



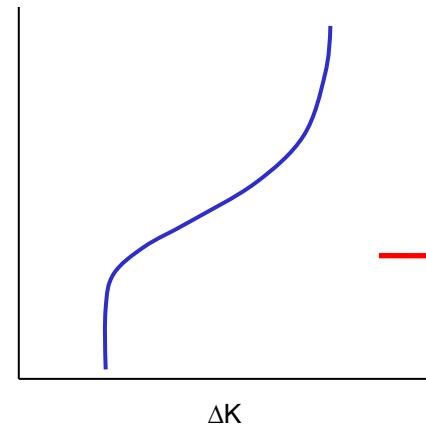
Loading



LEFM  
Solutions

$$\Delta K = \beta \Delta \sigma \sqrt{\pi a}$$

Material Data

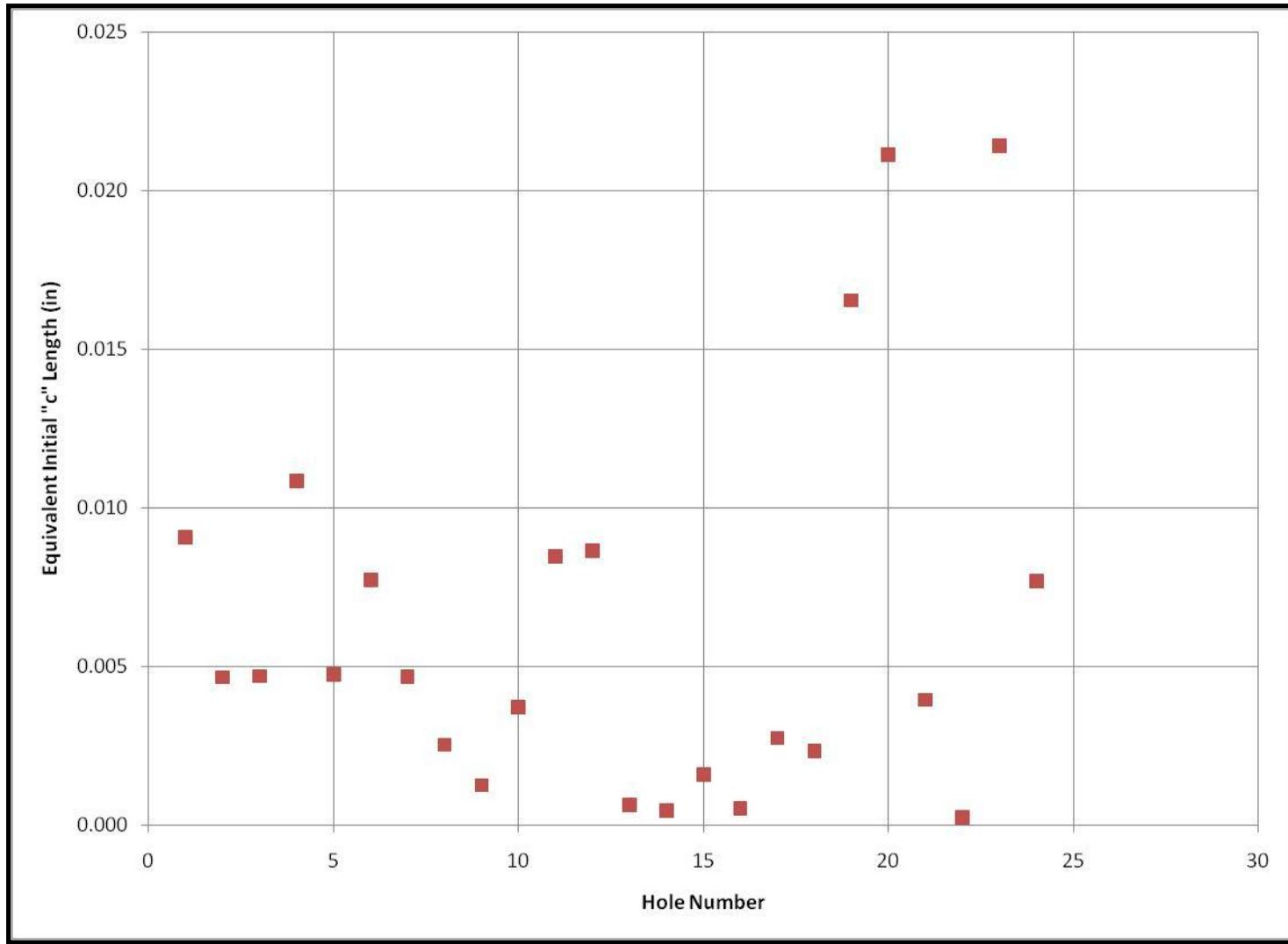


EIFS

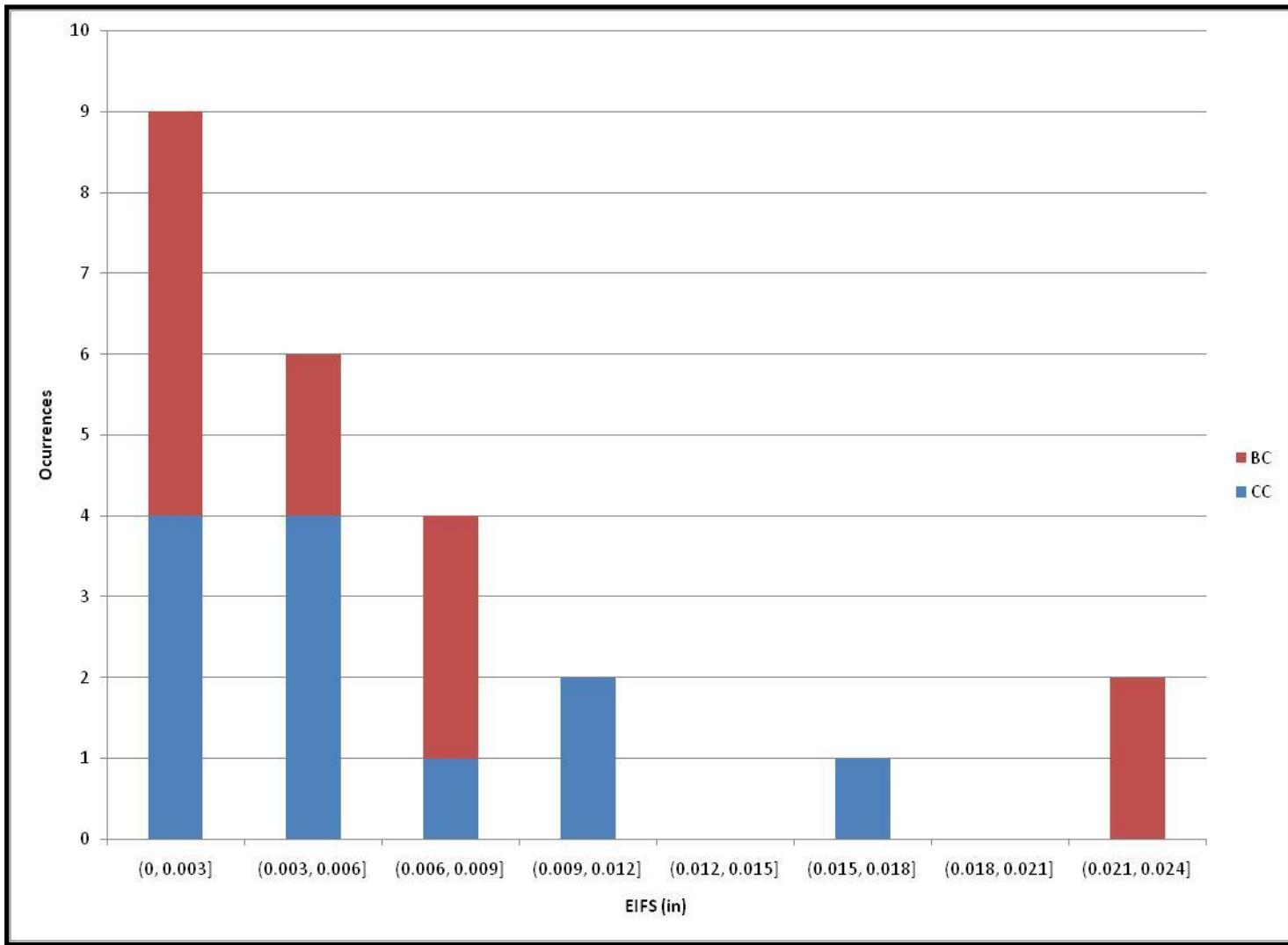
# Example User Input

Model	Hole Diameter	Thickness	Width	"a/c" ratio	SMF	SOLR	Material	Final "c" Length	Number of Cycles	Tolerance
1030	0.259	0.165	10	0.60	25.4	2.7	M7HG11AB1	0.025	530,318	53
2020	0.198	0.163	10	1.78	25.4	2.7	M7HG11AB1	0.045	569,312	57
1050	0.196	0.251	10	1.29	25.4	2.7	M7HG11AB1	0.017	569,312	57
1040	0.234	0.248	10	1.33	25.4	3.3	M7TF11AB1	0.045	584,910	58
1060	0.187	0.277	10	1.22	25.4	3.3	M7TF11AB1	0.056	577,111	58
2030	0.198	0.266	10	1.14	25.4	2.7	M7HG11AB1	0.055	577,111	58
1030	0.236	0.259	10	0.78	25.4	3.3	M7TF11AB1	0.023	522,520	52

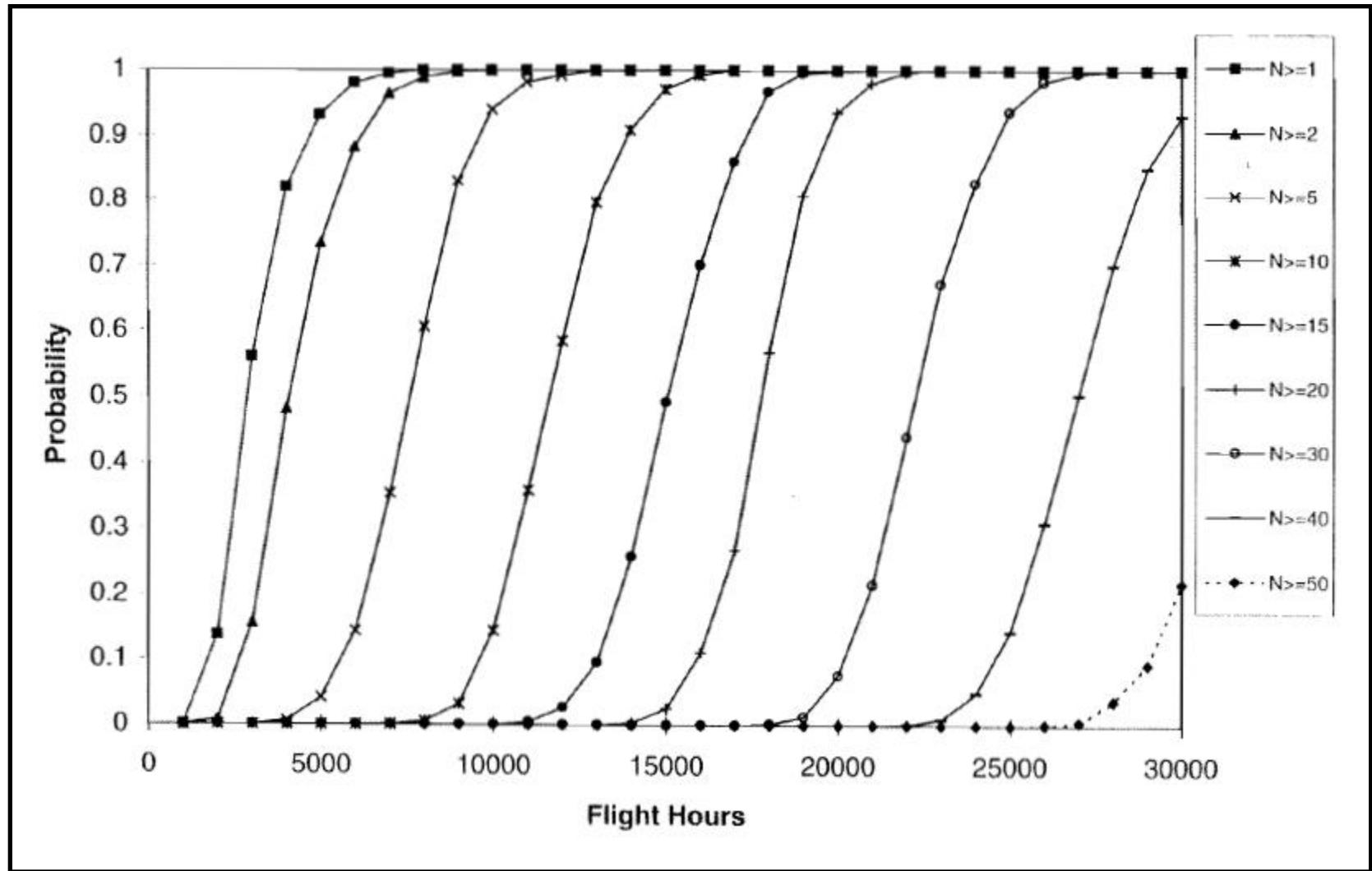
# Example Data



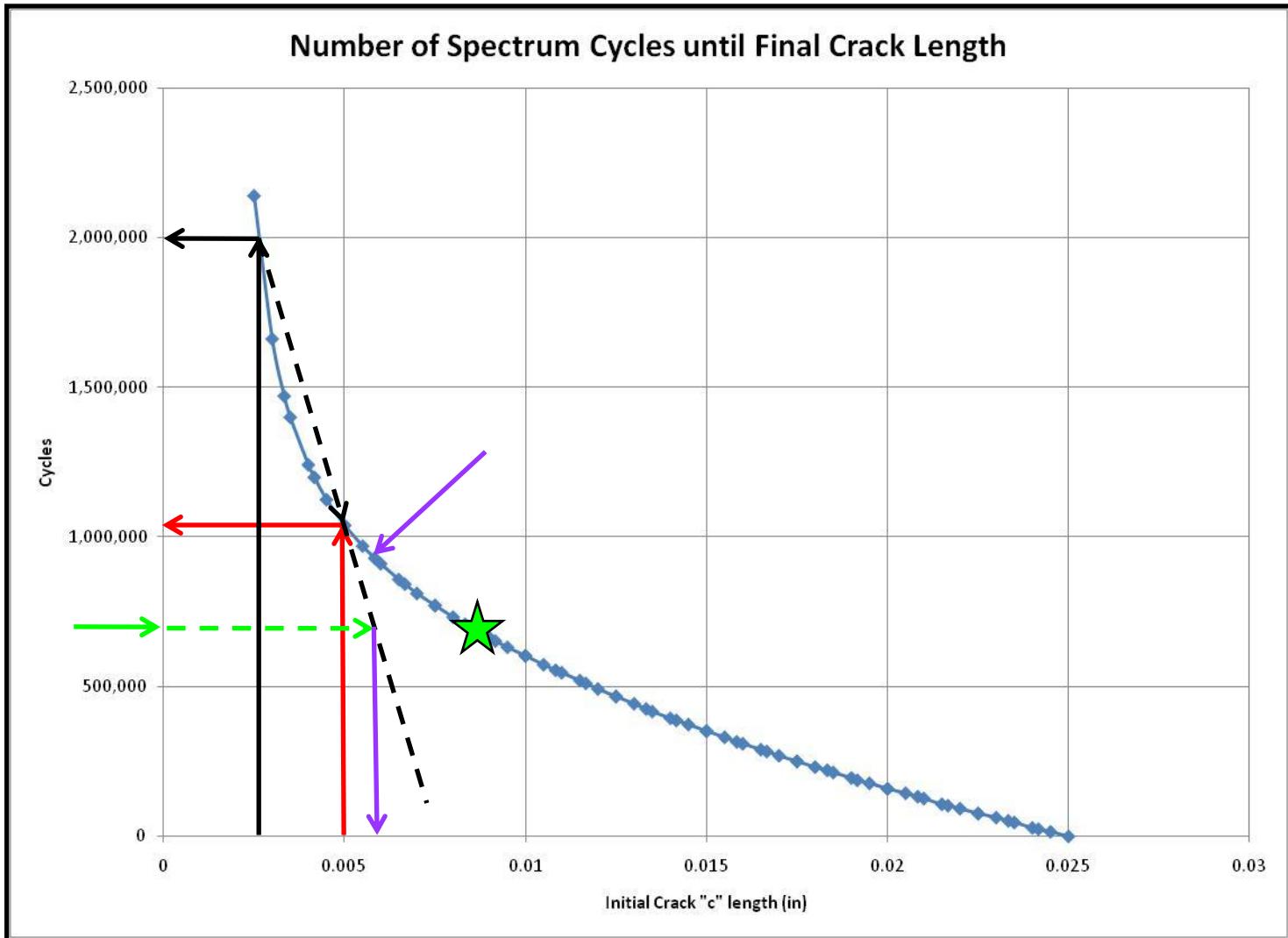
# EIFS Histogram



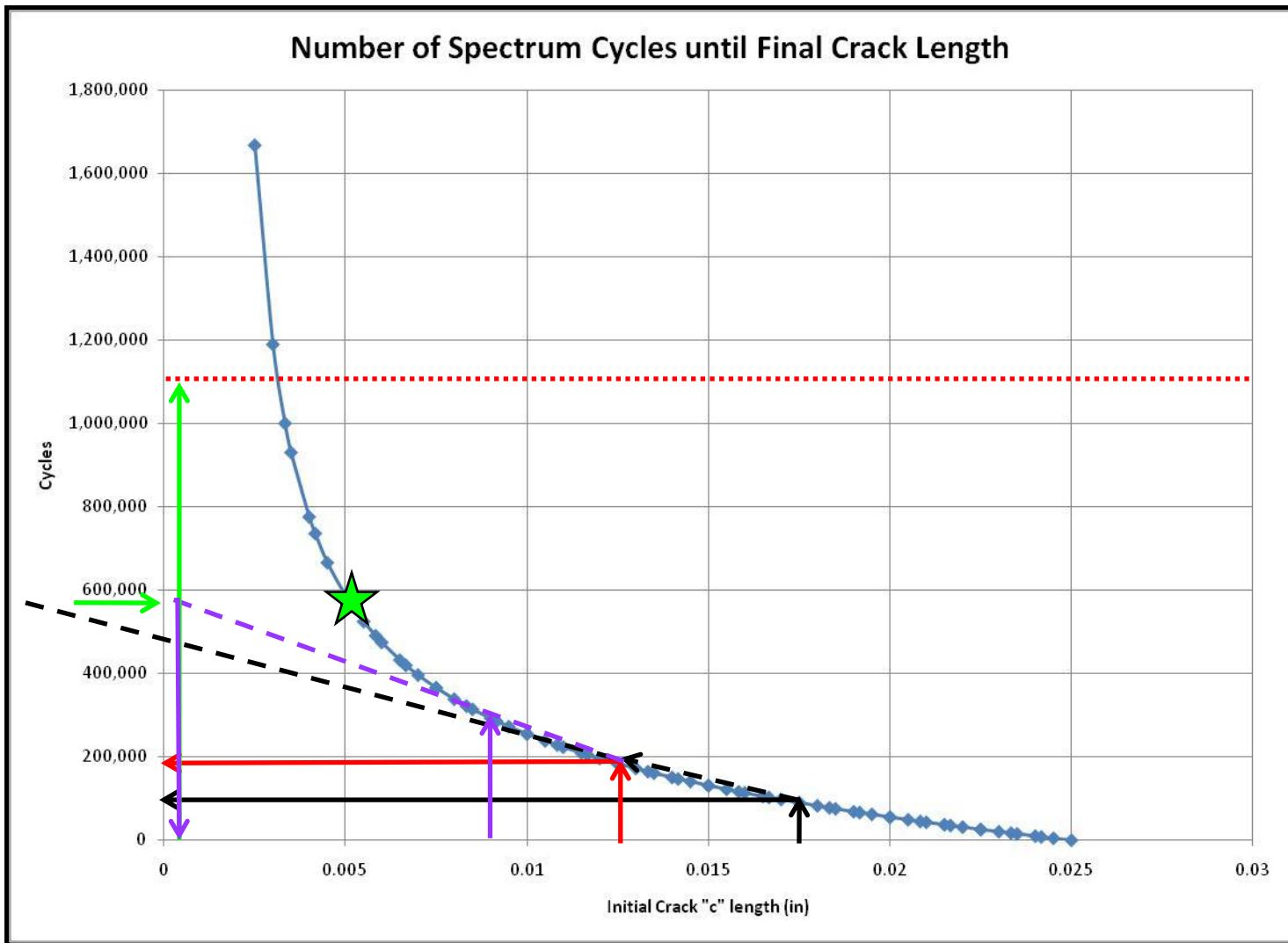
# Crack Length Exceedance



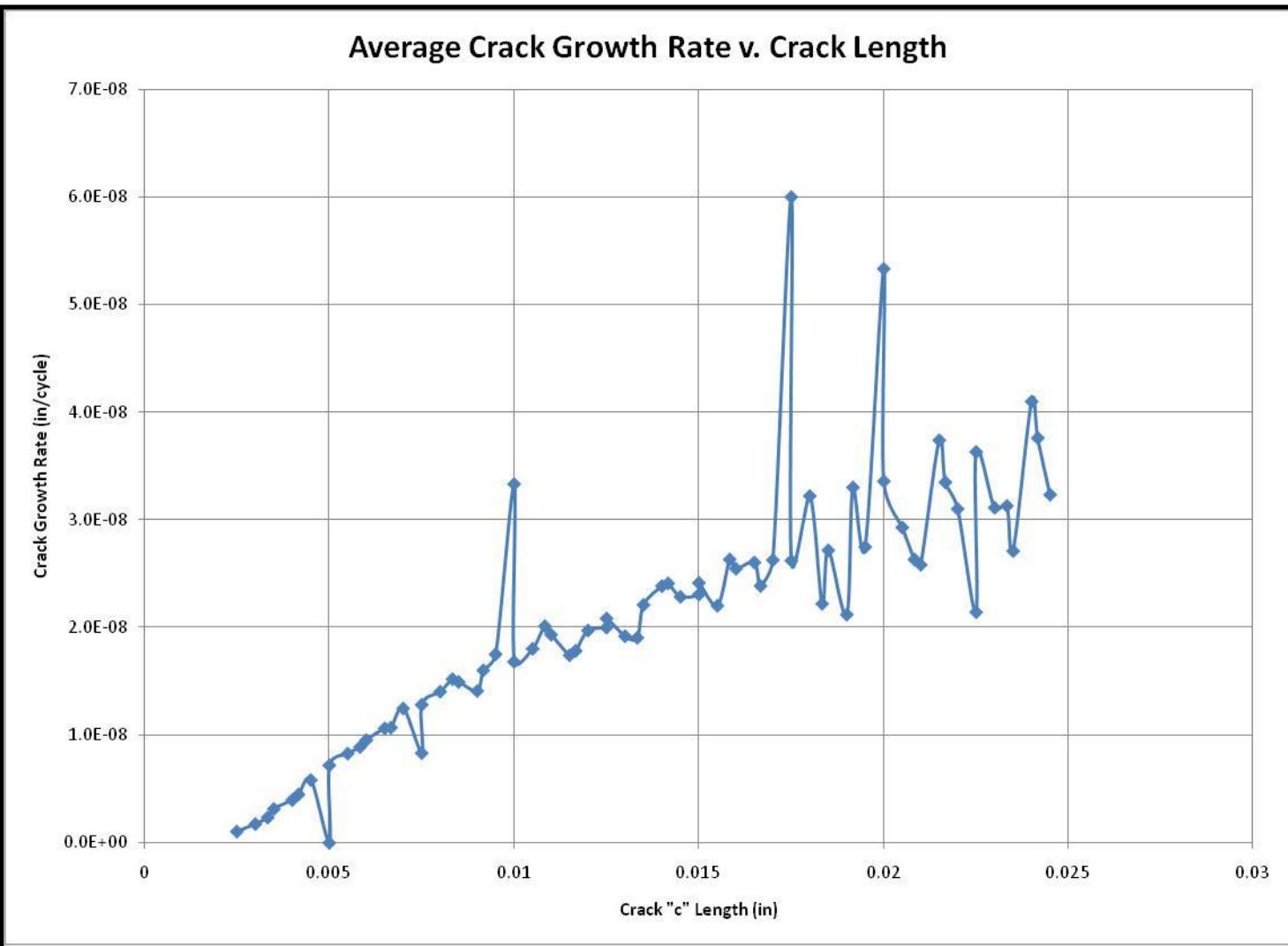
# EIFS Updated Guess



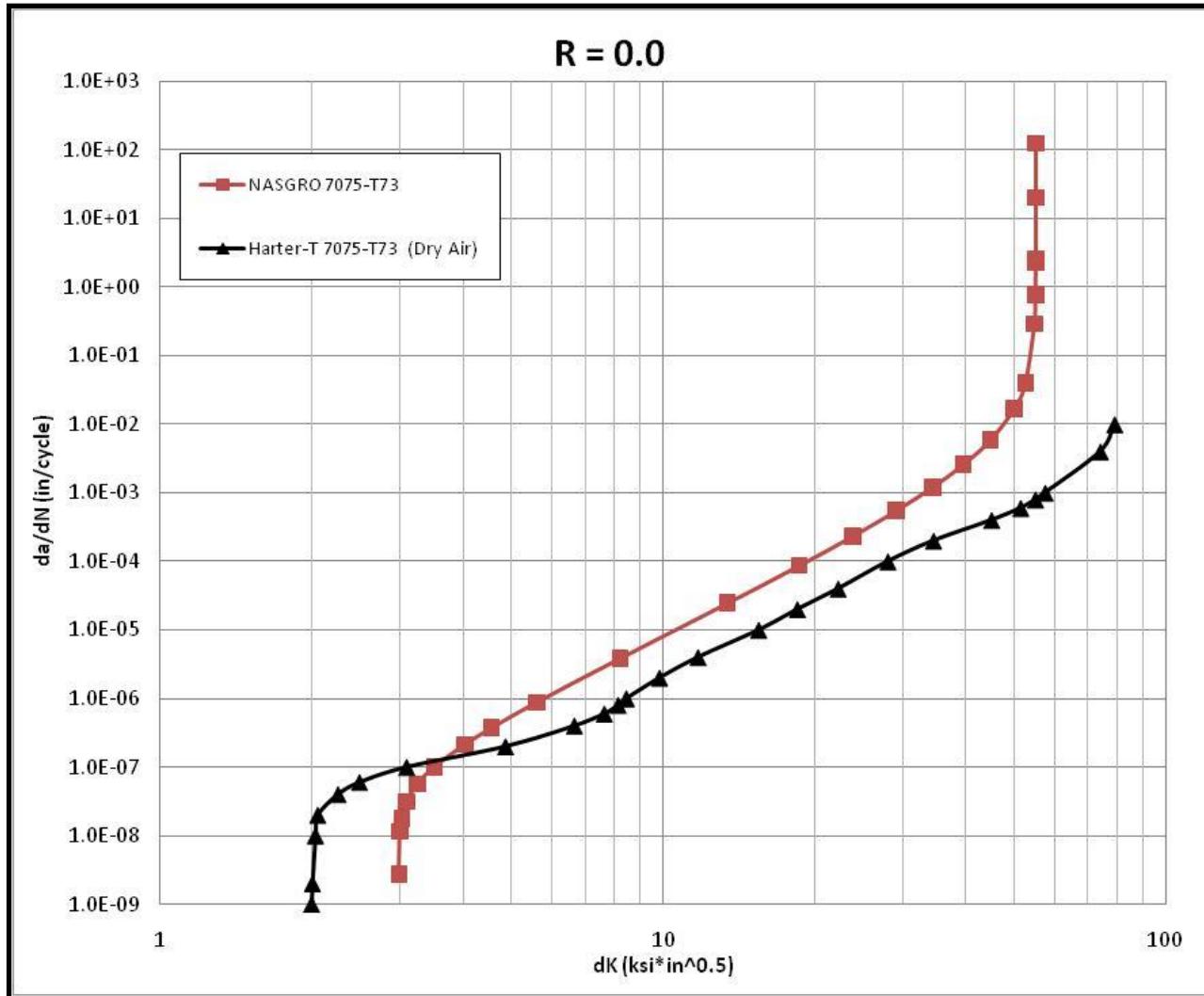
# EIFS Updated Guess



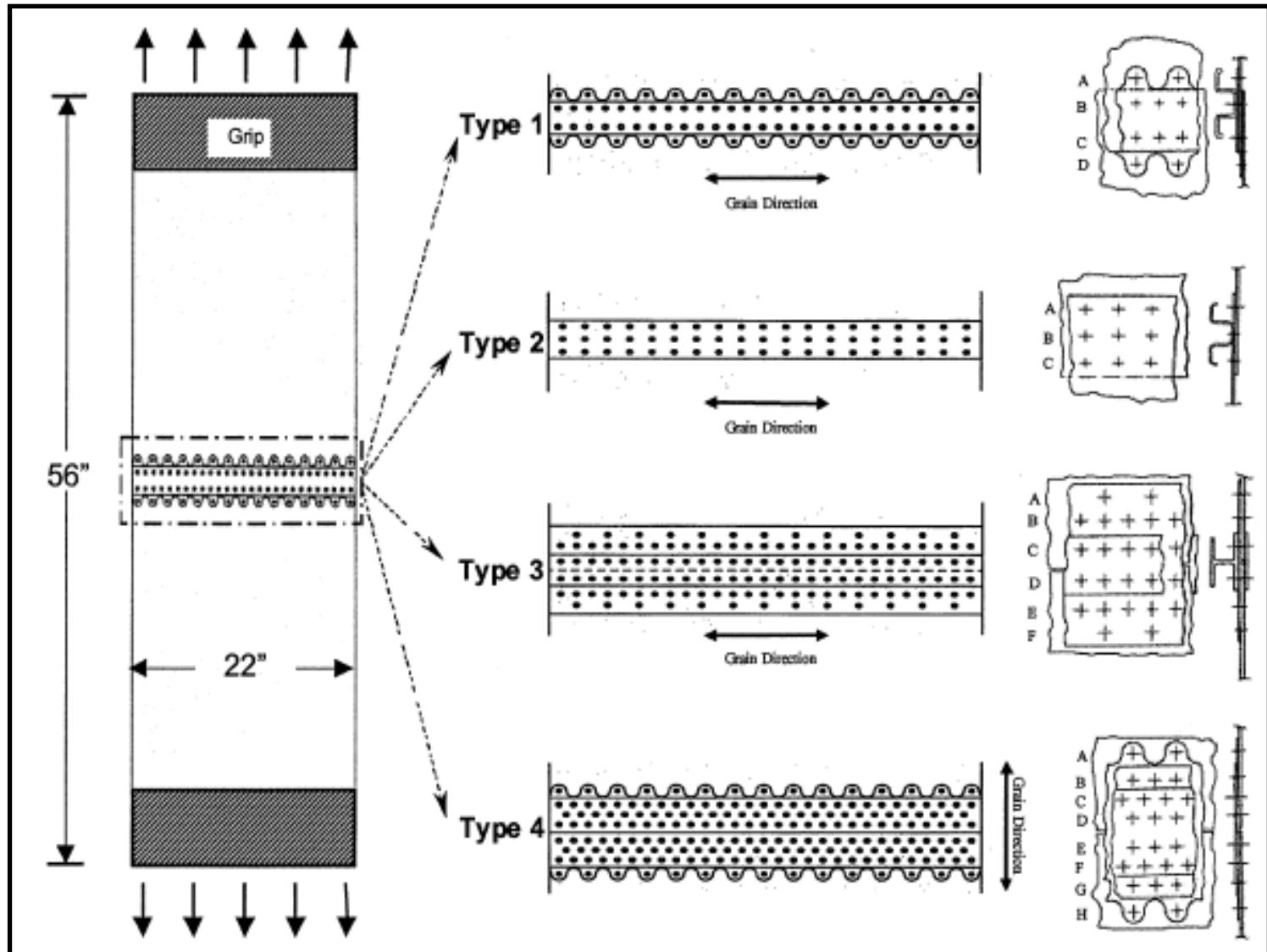
# EIFS Updated Guess



# Valid Assumptions?



# Valid Assumptions?



# Conclusions

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- AFGROW lends itself well to iterative LEFM and crack growth analyses via the COM option
  
  - The assumptions made for the analyses can have a significant impact upon the final “answer”
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# Questions?

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Valdez

